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AIR FORCE COMMUNICATIONS SERVICE

TRACALS EVALUATION REPORT

COMMUNICATIONS STATION EVALUATION REPORT

Ellsworth AFB, South Dakota

78/66C-122

27 March - 4 April 1978

AD NO.





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78/66C-122

27 March - 4 April 1978

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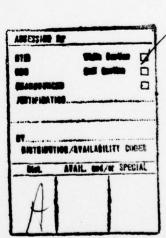
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Block 20 ABSTRACT

common problems with most other systems, include system alignment procedures, and antenna maintenance problems. The data and diagrams in this report can be used as a guide for anticipated performance of the communications system at Ellsworth AFB until there is a deletion, addition, relocation of equipment, or a change in horizon profile which would effect the system. This report will remain valid after relocation of the transmitter site.

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G-2-1/2	REFRACTIVE THEORY AND DEFINITIONS

SUMMARY

1-1. Ground/Air/Ground Communications System

- a. Ground-to-Air Communications: Ground-to-air coverage extends well beyond the limits of the Ellsworth AFB air traffic control (ATC) mission area. Coverage is adequate at all minimum vectoring altitudes (MVA) within the Ellsworth Control Area. The transmitter facility is scheduled to be relocated to a site near the receiver site. To gain an accurate picture of the coverage to be expected after the relocation, airborne checks were done from the receiver site using a spare UHF transmitter. The data collected illustrates that communications coverage will be adequate after relocation of the transmitter facility.
- b. Air-to-Ground Communications: Air-to-ground communications was adequate for the Ellsworth ATC mission. The coverage exceeded predicted values for all quadrants.
- c. Audio System Alignment: A high noise level in the air traffic controllers' headsets was present when the input to the regulated output amplifiers (ROA) was set for -35 dBm. When the input was set for -10 dBm the noise diminished. A local telephone company compression amplifier is located between the console and the ROA. This device amplifies the noise generated in the RAPCON to a level within the input range of the ROA. Setting the ROA input to -10 dBm reduces this problem and improves the quality of communications.
- d. Radio Frequency Interference (RFI) at Receiver Site: Numerous low frequency signals (below one MHz) were measured in the receiver building. These signals contributed to high noise levels measured in the receiver when the RF input was low. When the RF input is -90 dBm or greater, the noise diminishes.
- e. RAPCON Backup Radio Antenna Tower: The RAPCON backup radio antenna tower causes one degree of screening for the local weather radar. A proposed lowering of the tower 20 feet to eliminate the radar screening will significantly increase the radio line-of-sight screening for the approach end of runway 13 and will cause a decrease in coverage for that area.
- f. AT-197/GR Antenna: During the evaluation, the antenna in use was an AT-197/GR antenna. This antenna provides increased coverage at low takeoff angles compared to the AS-1097/GR antenna. At the altitudes flown during the evaluation, the AT-197/GR has 11 dB more gain than the AS-1097/GR.
- 1-2. <u>Power Systems</u>: Primary and backup power systems for all facilities were adequate and reliable.

2. RECOMMENDATIONS

2-1. Ground/Air/Ground Communications System

- a. Recommend the AT-197/GR antenna be utilized rather than the AS-1097/GR (see para 4-5a).
- b. Recommend the input to the ROA be tested at -10 dBm rather than -35 dBm (see para 4-5c).
- c. Recommend that maintenance personnel use the evaluation team's system alignment procedures until the technical order change is received (see para 4-5c).
- d. Recommend that an RFI team be requested to investigate the source of the low frequencies causing noise in the receiver site (see para 4-5d).
- e. Recommend that careful consideration be given to the increased screening resulting from the proposed lowering of the RAPCON backup radio antenna tower (see para 4-5e).
 - 2-2. Power Systems: No recommendations.

3. GENERAL INFORMATION

3-1. Facility Data

a. General

Location: Ellsworth AFB, South Dakota Communications Area: SACCA Unit: 2148 Communications Squadron

Evaluation Period: 28 March - 4 April 1978

b. Communications

Control Tower Coordinates: 44° 08' 44" N
103° 05' 39" W
Control Tower Site Elevation: 3212 feet MSL
RAPCON Coordinates: 44° 08' 44" N
103° 05' 35" W
RAPCON Site Elevation: 3212 feet MSL
Transmitter Site Coordinates: 44° 08' 41" N
103° 07' 39" W
Transmitter Site Elevation: 3235 feet MSL
Receiver Site Coordinates: 44° 08' 01" N
103° 06' 02" W
Receiver Site Elevation: 3194 feet MSL
Receiver Antenna Height: 80.5 feet AGL

3-2. Runway Data

Airfield Coordinates: 44° 08' 30" N 103° 06' 30" W Airfield Elevation: 3276 feet MSL Magnetic Declination: 13° E

3-3. Mission Area

- a. The Ellsworth Terminal Control Area is that airspace extending from the surface up to and including 12,000 feet MSL within a 40 NM radius of the Ellsworth TACAN (see TAB A-1).
- b. The Ellsworth Control Zone is the controlled airspace which extends from the surface up to and including 12,000 feet MSL. Its boundaries are described as a five statute mile radius of the geographic center of Ellsworth AFB with a 2.5 mile extension on either side of the Ellsworth TACAN 116 radial extending to 7 miles and 8 miles respectively. These extentions are needed to provide controlled airspace to include instrument approach and departure paths.
- 3-4. <u>Mission Responsibility:</u> The Ellsworth RAPCON is responsible for providing terminal air traffic service within its area of control, which includes Rapid City Regional Airport and many other small airports. The Ellsworth AFB Control Tower is responsible for providing control of visual flight rules (VFR) air traffic in its control area. Flying at Ellsworth AFB is moderate to heavy.

3-5. Primary Using Agency/Aircraft Supported: The primary operational user of Eilsworth AFB is the 28 Bombardment Wing flying B-52s. Many types of commercial and military aircraft use the base.

3-6. ATC Facilities

a. RAPCON

- (1) AN/FPN-47 Airport Surveillance Radar
- (2) AN/TPX-42 Air Traffic Control Radar Beacon System
- (3) AN/FPN-16 Precision Approach Radar
- (4) Four Channel Communications Control System/301A Key System

b. Control Tower

- (1) AN/GSA-135 Console
- (2) Four Channel Communications Control System/302A Key System
- (3) Bright Radar Indicator Tower Equipment

c. NAVAIDS

- (1) AN/GRN-27 Solid State Instrument Landing System
- (2) AN/GRN-19A Tactical Air Navigation
- 3-7. <u>Logistic Support:</u> Logistic and precision measuring equipment laboratory support is provided by the host base.

3-8. Key Personnel

a. Ground Evaluation Personnel

1Lt A.C. Mathews, Team Chief/Electrical Engineer MSgt W.V. Rogers, NCOIC, TRACALS Evaluation Team SSgt P.A. Tovar, TRACALS Communications Evaluator Sgt R.J. Herrera, TRACALS Communication Evaluator SrA S.F. Jurash, Geodetic Surveyor

b. Airborne Evaluation Personnel

Capt L.R. Duncan, Aircraft Commander

Capt J.E. Lawrence, Pilot

MSgt F.F. MacMahon, Flight Inspection Technician

SSgt M.J. Aufieri, Flight Inspection Technician

TSgt F.H. Hutchinson, Jr., Flight Mechanic

c. Facility Personnel Contacted

LtCol J.W. Cowan, Commander
Capt. S.J. Geertz, Chief of Maintenance
Capt. J.J. Hayden, Chief, ATC Operations
1Lt B.P. Heseltine, Chief Controller, RAPCON
CMSgt J.L. Mallard, Chief Controller, Control Tower
SMSgt D.W. Schroeder, Maintenance Superintendent
MSgt J.W. Knight, NCOIC Radio Maintenance
SSgt J.M. Nielson, NCOIC Radar Maintenance

4. GROUND/AIR/GROUND COMMUNICATIONS SYSTEM

4-1. System Description: Air traffic control communications at Ellsworth AFB are provided by remotely controlled VHF/UHF radio equipment. Landlines interconnect the remote transmit and receive radio facilities with the RAPCON and control tower. A four channel communications control system is used to provide keying, amplification and control of the transmit and receive audio signals to and from the remote facilities. The communications antennas are mounted approximately 80 feet above the ground on steel towers. Each control facility has its own backup radio equipment.

VHF/UHF Radio Equipment	<u>Qnty</u>	Freq (MHz)	<u>Use</u>
AN/GRT-21 Transmitter AN/GRT-22 Transmitter AN/GRR-24 Receiver AN/GRR-25 Receiver AN/GRC-171 Transceiver AN/GRC-175 Transceiver	10 19 19 10 2 2	119.5 121.1 121.5 125.3 126.2 134.1 236.6 243.0 253.5 259.1 271.3 272.2 275.8 284.0 289.4 363.8 375.2 390.8 396.0	RAPCON Primary Clearance Delivery Emergency RAPCON Discrete Control Tower Primary RAPCON Discrete Control Tower Discrete Emergency Control Tower Primary RAPCON Discrete RAPCON Primary RAPCON Discrete Control Tower Ground Control RAPCON Discrete Clearance Delivery RAPCON Discrete
Ancillary Equipment	Qnty	<u>Use</u>	
CU-547/GR AN/GSA-135 Four Channel/302A Key System Four Channel/301A Key System AT-197/GR AS-1097/GR AS-1181/UR	5 1 1 1 17 2 16	Antenna Coupler Control Tower Console Control Tower RAPCON UHF Antenna UHF Antenna VHF Antenna	

4-2. Equipment Status

a. Facility Equipment Status: Equipment checks were accomplished using procedures described in the equipment technical orders. Where no procedures are given, AFCSP 100-61, Vol XIII was used as a guideline to ascertain the operational status of the equipment. The equipment specifications and test results are shown in TABs E-1-1 thru E-5-4.

- (1) Transmitter Site: Four of the transmitters checked had high output power and two had modulation in excess of 100%. One transmitter checked had high distortion which was corrected by adjustment. All of the ports on one antenna coupler had high insertion loss which could not be corrected externally.
- (2) Receiver Site: Three of the receivers checked had incorrect signal-to-noise ratios. All but one of the high noise conditions were corrected by receiver alignment. The other receiver would not tune to specifications. One receiver had a low audio output which was corrected. Only one antenna was found to be defective during the evaluation and this antenna was replaced by the squadron antenna team. One spare antenna cable had no markings. Investigation revealed that no antenna was connected to the cable.

4-3. Environmental Factors

a. Siting Characteristics:

- (1) General: Ellsworth AFB is located on the extreme western side of the state of South Dakota about 10 miles east of Rapid City. The Black Hills are approximately 25 miles to the west and form the only significant screening in the area. The terrain to the south, east and north is generally rolling prairie.
- (2) Transmitter Site: The transmitter site is located southeast of the approach end of runway 13. The terrain surrounding the site is primarily flat, treeless pasture land.
- (3) Receiver Site: The receiver site is located northeast of the approach end of runway 31 on terrain identical to that surrounding the transmitter site.

b. Weather

(1) Surface Climatology: The climate of Ellsworth AFB is characterized by short summers and long, moderately cold winters. Cold waves usually occur about five times each winter with varying severity, but they are not prolonged as a rule. They are often broken up by warm and pleasant weather, usually brought on by the onset of chinook winds Precipitation amounts to about 16 inches annually, while the mean annual snowfall is 38 inches. Snow has occurred in every month except July and August.

(2) Propagation Climatology:

(a) During the winter Ellsworth AFB is primarily dominated by continental polar (cP) air masses. This cP air often moves upslope into the area around the southwest periphery of the Canadian cP high pressure outbreaks. In the early part of the season, before the

ground is snow covered, slightly subrefractive conditions will exist during daylight hours, while at night conditions will become standard to slightly superrefractive by early morning. Later in the season, as snow covers more of the surface, conditions become more conservative, remaining mostly standard to slightly subrefractive. However, on nights with clear skies and calm winds, radiational cooling will create a low level temperature inversion and moderate superrefractive conditions will occur. Strong surface winds that develop shortly after sunrise and are predominant in this area, cause rapid vertical mixing to moderate levels and destroy surface and low level ducts that may have been formed by nocturnal cooling. Also, during this season the warm chinook winds pick up moisture in the lower levels from melting snow creating conditions favorable for ducting. Due to surface mixing and turbulence, these ducting layers which are often strong, are formed at the top of the convective levels near 1000 to 2000 feet. These ducts will persist for the duration of the chinook. Maritime polar (mP) air moving into this area from the Pacific has been dried out and warmed in coming over the mountains. It is warm compared to the cP air present in the area and is forced aloft over the cooler cP air. A subrefractive layer is usually formed at the mixing zone of these two air masses.

- (b) During the summer months maritime polar (mP) air is the predominant air mass in this area. Although normal propagation can be expected to occur the majority of the time, this is also the season of maximum superrefractive activity. The maritime polar (mP) highs moving in from the Pacific frequently have inversions, due to subsidence, up to 10,000 feet. The effects of chinook winds form inversions 1000 to 2000 feet above the surface. These inversions are favorable to the formation of superrefractive layers. Often nocturnal cooling creates favorable conditions for the formation of moderate to strong superrefractive layers near the surface. These conditions will persist until some change of weather regime occurs.
- (c) In the spring and fall propagation conditions become standard to slightly subrefractive. These seasons are transitional periods between the more frequent superrefractive conditions of summer and the standard to subrefractive conditions of winter.
- (d) The charts "Frequency of Refractive Conditions in Percent" (TAB G-1) are derived from summaries of atmospheric refractive indexes prepared by the USAF Environmental Technical Applications Center (AWS). They are computed for the nearest rawinsonde station considered to be representative of this site. The charts represent a count by month, over the period of record of three or more years, of the minimum gradient category in percent frequency of occurrence. Only the one minimum gradient category in each upper air sounding has been counted. For this reason subrefraction is seldom shown on the charts, as more negative gradients will usally be found and counted. A discussion of refractive theory, and a description of the refractive index categories and their corresponding gradients in N-units per 1000 feet is found in TABs G-2-1/2.

- (3) Evaluation Weather Conditions: Weather did not adversely affect the airborne data collection phase of the evaluation. During the flying days there was good mixing action in the atmosphere resulting in ideal electromagnetic propagation conditions.
- 4-4. Evaluation Profile: The overall objectives of the evaluation were to define the capabilities and limitations of the air traffic control communications equipment in the installed environment and to optimize the performance of the system. These objectives were met by making the siting and environmental studies discussed in paragraph 4-3 and performing the equipment, system, and airborne checks described below.
- a. Ground Tests: Ground tests were performed prior to the airborne tests. They consisted of two types: equipment checks and loop tests.
- (1) Equipment Checks: Equipment checks were performed prior to loop and airborne tests to ensure proper operation of major end items. The results of the checks were compared with technical order specifications. Where technical order specifications were not listed, the data base built from prior evaluations was used as a reference in determining equipment performance. Additional information, such as antenna placement measurements (TABs D-1-1/2), was also obtained. Adjustments of equipment for optimum operation were made immediately, if possible without extensive maintenance. Other problems were identified to maintenance personnel for correction. The corrected readings are included in the "adjusted" column of the equipment check forms. The audio amplifier measurements were recorded after the amplifiers were adjusted for normal operation.
- (2) Loop Tests: Loop tests were utilized to evaluate the system performance of the previously tested end items. An operational position in the RAPCON and a maintenance position in the control tower were used for the loop tests. A one kHz tone was injected into the microphone amplifiers for simulation of a normal voice input. One frequency at a time was keyed. The signal levels, signal-to-noise, and modulation measurements were taken on the transmit portion of the system with a dummy load placed on the transmitter. The one kHz tone was removed and noise and carrier power measurements were taken. The audio measurements were taken on the receiver side of the system using a 30 percent modulated RF carrier connected to the input of the receiver equipment. The audio measurements were taken at accessible points in the system. The resulting data were used to determine the signal levels presented on the Loop Test Line Level Diagrams (TABs E-3-4 and E-4-6).
- b. Airborne Tests: The airborne tests were accomplished using a C-140A flight inspection aircraft flying radials and orbits off the Ellsworth TACAN. The automatic gain control (AGC) current of the airborne receiver was used to obtain the receive signal level (RSL) of

the communications frequency under test. Fourteen radial tracks and one orbit were flown using the aircraft receiver to measure the ground-to-air transmit signal strength; eight radial tracks and one orbit were flown using a ground receiver to measure air-to-ground transmit signal strength (see TAB F-1). Radial track measurements were used to determine vertical radiation patterns, and orbital tracks were used to determine horizontal patterns. Prior to the airborne tests, the aircraft and ground receiver AGC currents were calibrated in dBm by injecting known signal levels into the receiver's RF transmission line and annotating the strip chart recordings. The ground transmitter was continuously keyed with the output power set at 10 watts. The aircraft transmitter output was measured and recorded. While measuring the air-to-ground signal strength the aircraft transmitter was keyed on and off at ten seconds intervals.

4-5. Analysis of Evaluation Results

- a. Ground-to-Air Communications: AFM 55-8 tolerances specify clear and readable communications at altitudes which meet operational requirements at a minimum distance of 15 NM for the control tower and 30 NM for approach control. Emergency communications is desired to extend as far as possible. Pilot-to-Forecaster communications is required to 100 NM at 20,000 feet above the site elevation (AWSR 105-12). The RSL measurement flight profile is shown in TAB F-1 and the results of the ground-to-air signal strength measurements are contained in TABs F-2-1/2. The data from the orbits and radial tracks give a composite three dimensional picture of the electromagnetic radiation pattern of the antennas under test.
- (1) Horizontal Radiation Pattern: The orbit was flown at a radius of 30 NM and an altitude of 9600 feet MSL. The RSL was relatively constant throughout the orbit. This was due to the flat, treeless, uniform terrain encompassing the base. The mean RSL was -75 dBm utilizing an AT-197/GR antenna. This value well exceeds the predicted value of -93.2 dBm. The predicted value is derived from theoretical calculations which utilize antenna gain at specific takeoff angles, frequency, transmission line losses, distance, and output power. Antenna gain values come from manufacturer's data for the AS-1097/GR antenna. Utilizing antenna gain values for the AT-197/GR antenna, the predicted RSL is -82.0 dBm. At the altitude and range for the orbits, the takeoff angle was 1.85 degrees. The AS-1097/GR antenna has a gain of -9 dBi at this takeoff angle, whereas the AT-197/GR has a gain of +2.15 dBi. The AS-1097/GR is the antenna in use at most ATC remoted transmitter and receiver sites and is in use at Ellsworth. The predicted RSL for an AS-1097/GR at the altitude flown and at 30 NM out would be 0.2 dBm below the squelch threshold of the aircraft's receiver. The calculations do not take into account multipath propagation, weather, screening, or other environmental factors. The improved antenna radiation pattern of the AT-197/GR, flat terrain (which induced multipath and presented no screening), and excellent weather conditions accounted for the better than predicted RSLs.

- (2) Vertical Radiation Pattern: TABs F-3-1/6 show that coverage extended beyond the Ellsworth Control Area, exceeding 50 NM in all quadrants at the MVA. Due to the AT-197/GR antenna radiation pattern and environmental factors, radial coverage is extended compared to that normally found at sites utilizing the AS-1097/GR antenna. The aircraft's nose high attitude in flight produces an approximate five dB difference between outbound and inbound tracks. The effects of multipath radiation were evident when analyzing the chart recording taken on the 050° radial. The signal level varied almost sinusoidally, indicating the addition and cancellation of the signal due to reflections (multipath). The signal path was directly across the runway, taxiways, and ramps on the 050° radial. These flat, concrete surfaces, when combined with the flat, treeless terrain of the base, are excellent reflective surfaces, thus producers of multipath radiation.
- (3) Screening Effects: The only significant screening at Ellsworth is the Black Hills to the west. There is no significant on-base screening of the communications system, except as noted for the RAPCON backup radios (see para 4-5e).
- (4) Relocation of Transmitter Site: Scheme action is anticipated to begin late this year to relocate the transmitter site. The new site will be in building 6925 which is approximately 100 yards from the present receiver site (building 6922). To provide data which will be useful in the future, a spare UHF transmitter was set up in the receiver site. This transmitter was utilized during the ground-to-air portion of the evaluation. No VHF transmitter was available, therefore no VHF tracks were flown. VHF coverage will extend significantly further than UHF coverage. The longer wavelength propagates further than the shorter wavelength of UHF. Due to the close proximity of the present receiver site to the new transmitter site, the use of a transmitter at the receiver site gave an accurate presentation of ground-to-air communications coverage after relocation.
- b. Air-to-Ground Communications: Air-to-ground coverage normally exceeds ground-to-air coverage. Radial data generally supports this, however, the mean RSL for the air-to-ground orbit was -82 dBm. This lower reading is due to the use of an antenna multicoupler with its 2 dB of loss, and other environmental factors. No coupler was in use during the ground-to-air phase of the evaluation. The -82 dBm mean RSL is 15.5 dB above the squelch threshold of the ground receiver. As a result, communications coverage extends beyond the limits of the Ellsworth Control Area.
- c. Audio System Alignment: Some alignment procedures utilized for the four channel communications control system in the RAPCON specify testing the ROA with an input of $-35~\mathrm{dBm}$. When the ROA is set for this input, the controllers experienced high noise in their headsets, making

communications difficult to impossible. A local telephone company (TELCO) compression amplifier is located before the ROA. This causes background noise and voice signals to arrive at the ROA at similar levels. Maintenance personnel are not able to adjust the TELCO's compression amplifier. If possible, the TELCO compression amplifier should be used solely as an amplifier, if amplification is felt necessary prior to the ROA. Another option would be to adjust the compression amplifier to eliminate background noise with an audio threshold adjustment, if one is present. Due to the presence of the TELCO compression amplifier, the ROA must be tested with an input of -10 dBm rather than other levels mentioned in some instructions. Change 5 to TO 31Z3-22O-6WC-1 will include alignment procedures for the four channel communications control system which will specify the input to the ROA be set at a level providing the best possible communications. Until receipt of this change, maintenance personnel can provide optimum communications utilizing the evaluation team's alignment procedures.

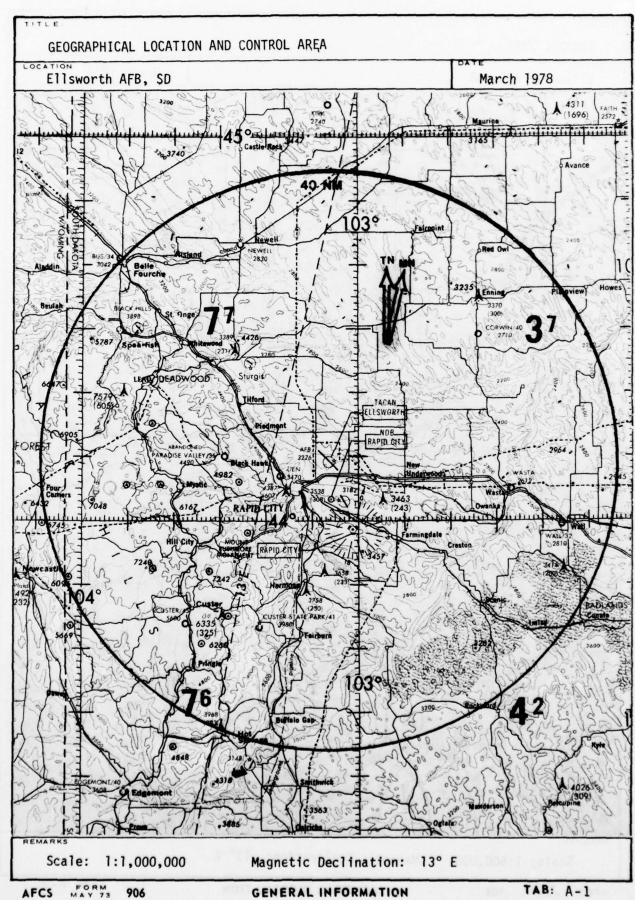
- d. Radio Frequency Interference at Receiver Site: High noise at the receivers' output was noted during the loop tests. This noise is evident when the RF input is at 3 uV (squelch threshold), and diminishes as the RF input to the receiver is increased. At the low input level internal receiver noise appears. As the RF signal level is increased the signal to-noise ratio increases. Investigation into the cause of this noise using a spectrum analyzer showed a number of distinquishable frequencies inside the receiver building. These frequencies were predominantly in the low frequency range (below one MHz). The data in TAB E-4-1 was taken when the RF input was greater than 3 uV which corresponds to -97.5 dBm at 50 ohms. Measured RSLs were significantly greater than -90 dBm, thus the RFI does not seriously effect communications, except when RSLs are very low.
- RAPCON Backup Radio Antenna Tower: The RAPCON backup radio antenna tower causes one degree of screening of the local weather radar pattern. The radar tower is approximately 50 yards south and at the same elevation as the radio tower. A scheme has been proposed to correct this by lowering the radio antenna tower 20 feet. Screening data (TAB B-2-4) shows that if the radio tower was lowered 20 feet, screening for the radios would increase. From 317° to 329° the screening angles would increase from +0.5° to +1.4°. Between 329° and 339° the screening angles would increase from 0° to +0.9°. The distance to the horizon would decrease from 15 miles to less than one mile. The Pride hangar (Building 7504) is the object causing this screening. The metal catwalk on the hangar's roof causes 0.5° of screening (from +0.9° to +1.4°). The approach end of runway 13 will be affected. Due to this screening angle an aircraft must be 150 feet off the ground for radio line-of-sight communications between 317° and 329° at one mile. The control tower (between 268° and 273°) will still be below the horizon and will not create additional screening for the runway if the antenna tower is lowered.

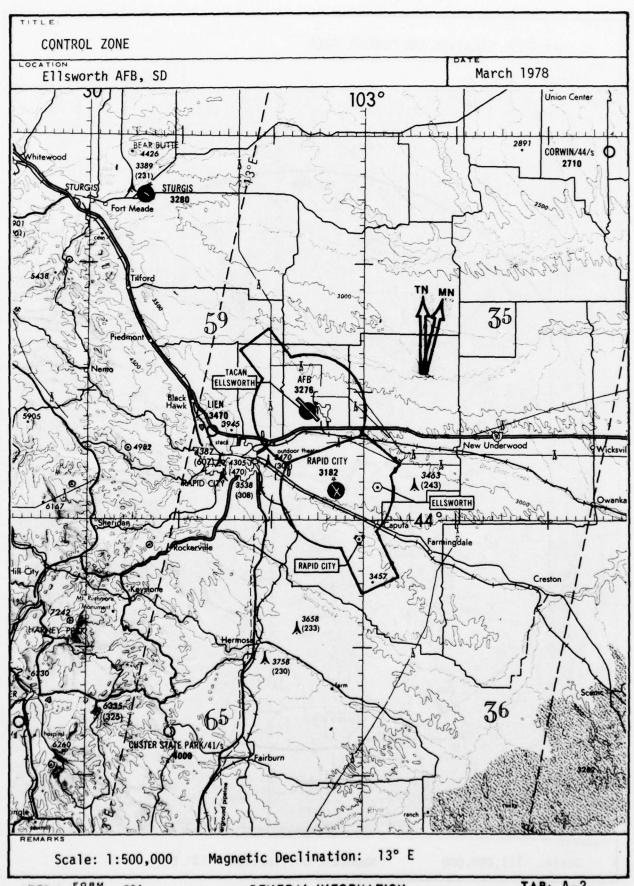
4-6. Capabilities and Limitations

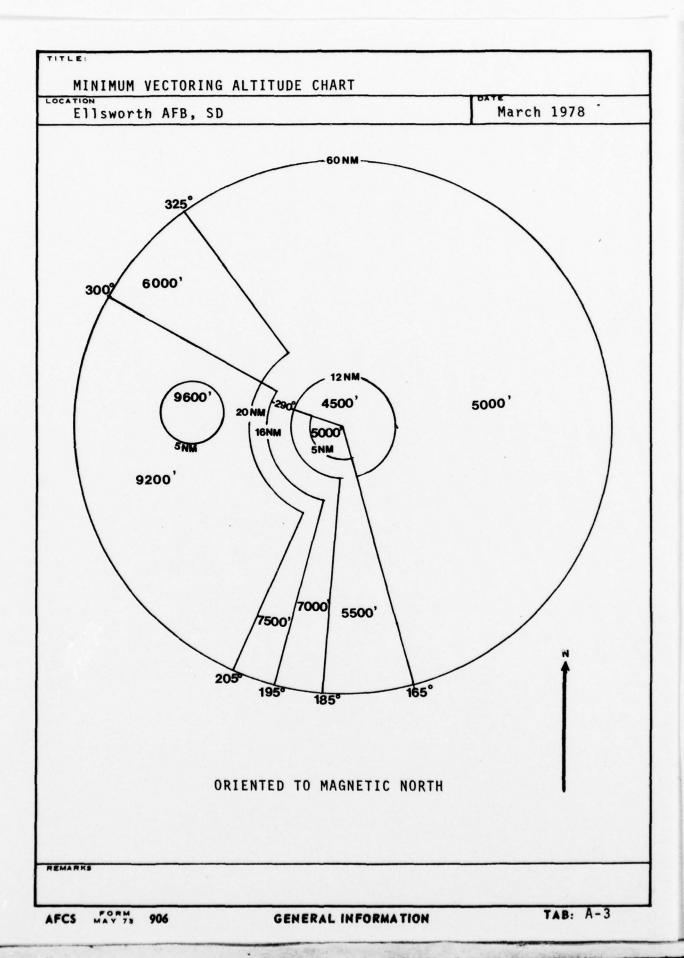
- a. Ground-to-Air Communication: Ground-to-air communications coverage extends beyond the limits of the Ellsworth Control Area at the MVA. Relocation of the transmitter site will not cause any decrease in coverage. Screening is not a factor. Multipath propagation is the only noticeable influence on coverage.
- b. Air-to-Ground Communication: Air-to-ground communications coverage extends beyond the Ellsworth Control Area at the MVA. The effects of multipath is the only influence noticed. Screening did not affect the communications coverage.
- c. Predictions: During normal operation, communications will continue to adequately serve the Ellsworth ATC mission area. The relocation of the transmitter site will ease maintenance and not affect the coverage. Environmental factors, weather and multipath, will influence the communications, but will not degrade it below desired levels, except under extraordinary conditions. Backup communications for the RAPCON will be affected by screening if the proposed lowering is accomplished. This screening will be evident on the approach end of runway 13.

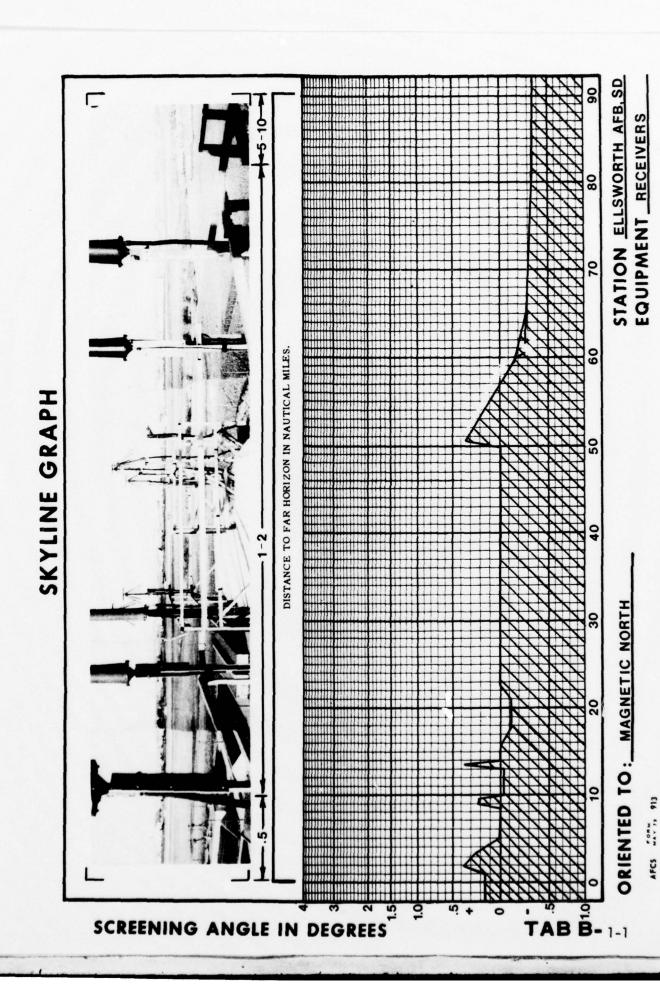
5. POWER FACILITIES

- 5-1. Equipment Details: Commercial power is provided for the RAPCON, control tower, transmitter, and receiver sites. Secondary power is provided by backup generators for each facility. A detailed list of the equipment and test results are presented in TABs E-5-1 thru E-5-4.
- 5-2. Equipment Status: The secondary power generators were in good operating condition. The automatic changeover unit at the transmitter site was awaiting parts. Manual changeover was accomplished and the generator performed satisfactorily.
- 5-3. Adequacy/Reliability: Primary and secondary power for all ATC communications facilities is adequate and reliable.





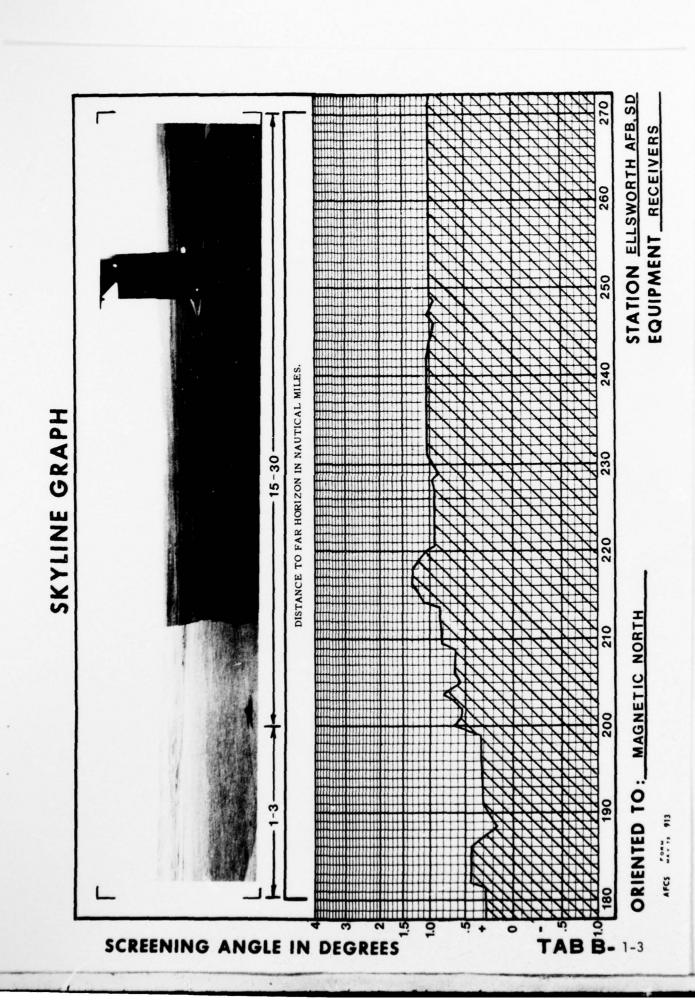


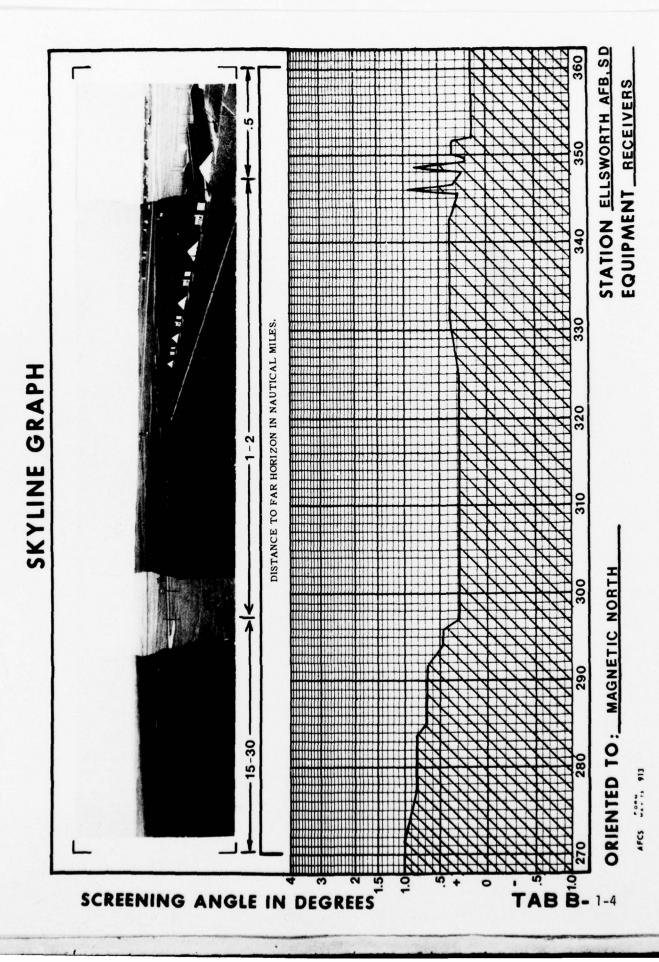


STATION ELLSWORTH AFB.SD EQUIPMENT RECEIVERS

ORIENTED TO: MAGNETIC NORTH

AFCS 913

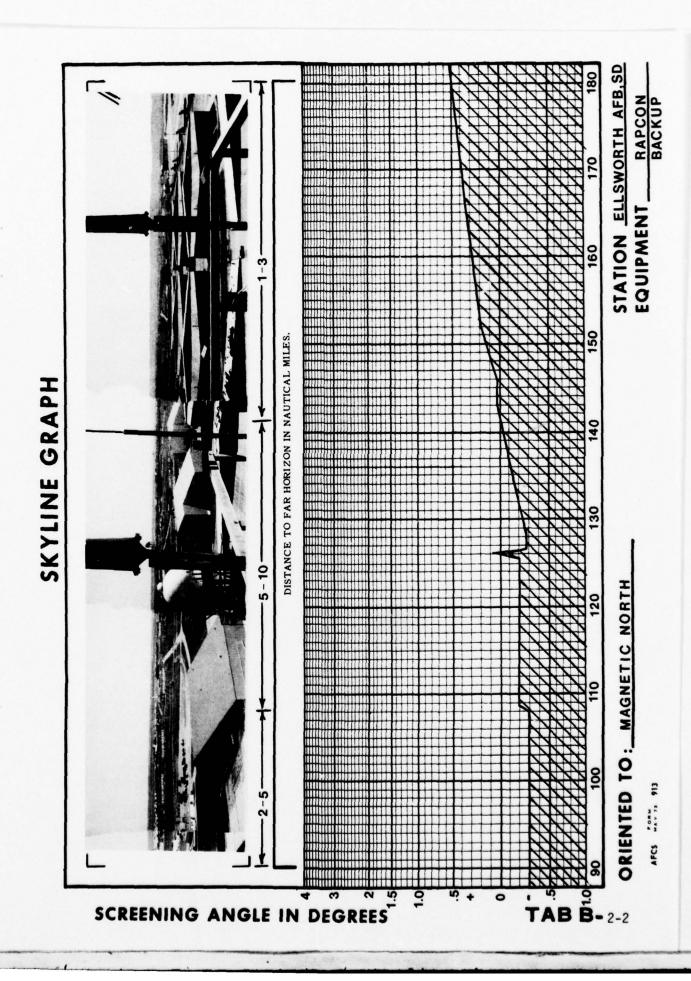




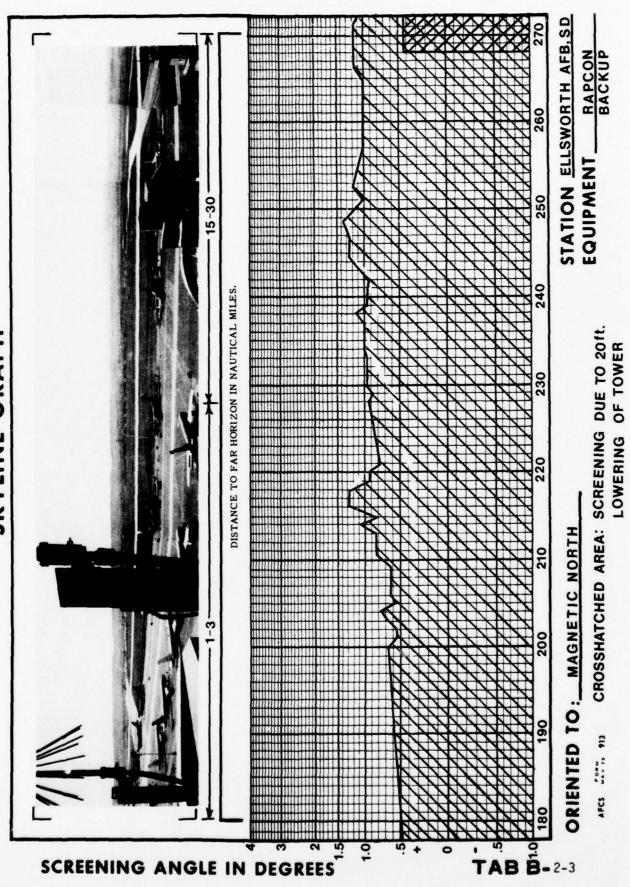
STATION ELLSWORTH AFB. SD EQUIPMENT RAPCON DISTANCE TO FAR HORIZON IN NAUTICAL MILES. SKYLINE GRAPH ORIENTED TO: MAGNETIC NORTH TAB B-2-1 SCREENING ANGLE IN DEGREES

RAPCON BACKUP

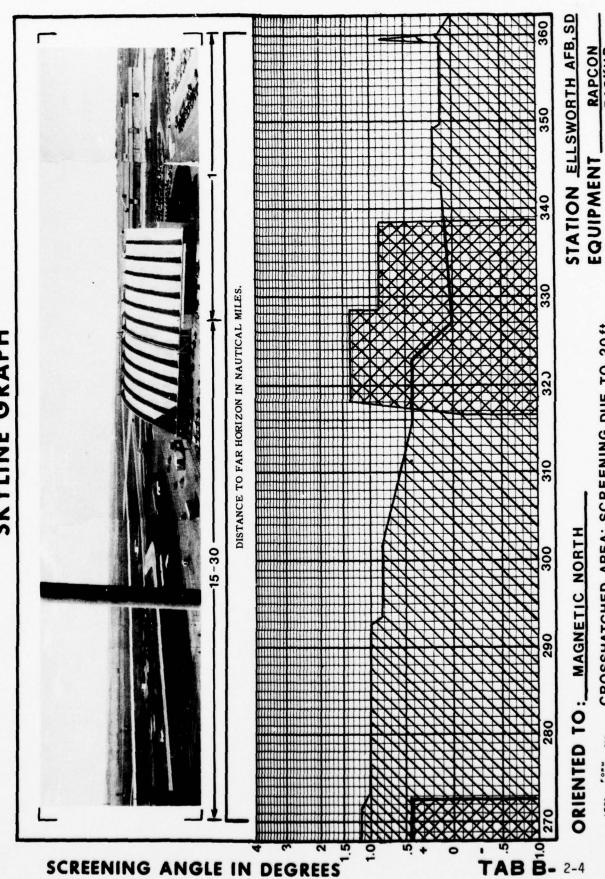
AFCS 913



SKYLINE GRAPH



SKYLINE GRAPH



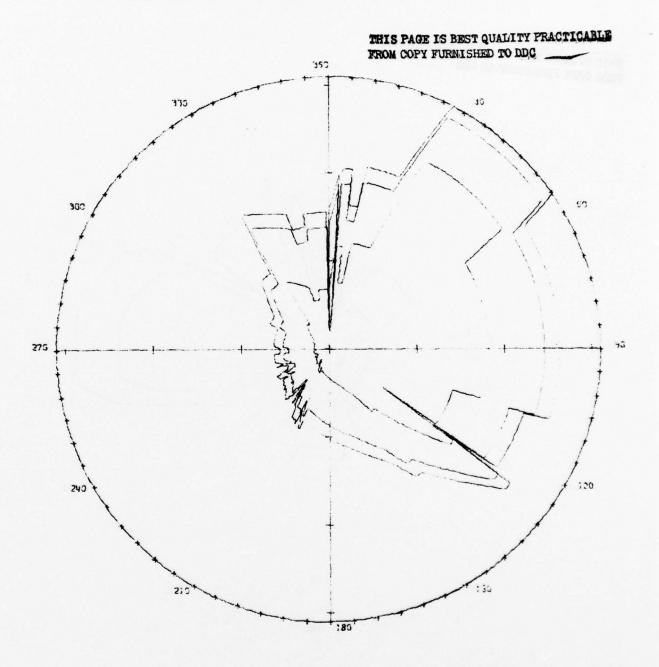
RAPCON

CROSSHATCHED AREA: SCREENING DUE TO 20ft.

AFCS ... 913

LOWERING OF TOWER

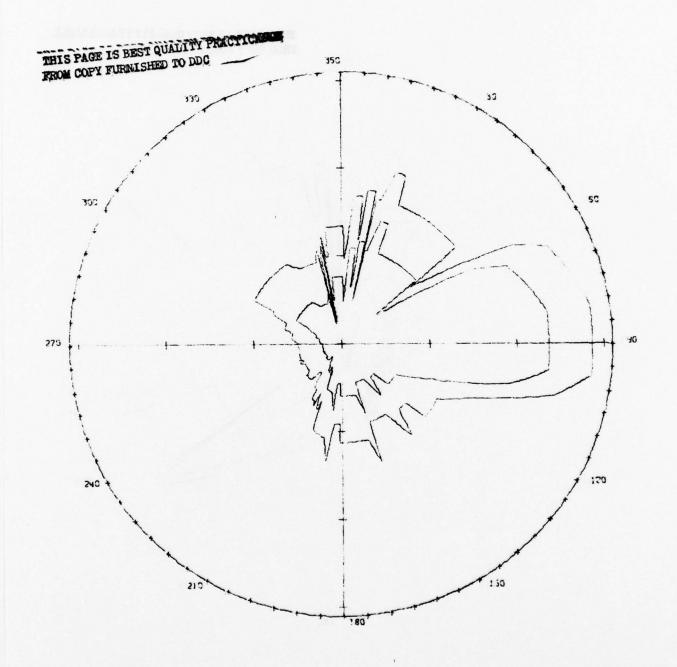
LINE OF SIGHT COVERAGE (RLS)



ELLSWORTH AFB RAPCON 29 MAR 78 ANTENNA ELEVATION 3317 FT MSL SCALE: 1 INCH = 35 NM ORIENTED TO MAGNETIC NORTH VARIATION 13 DEG E

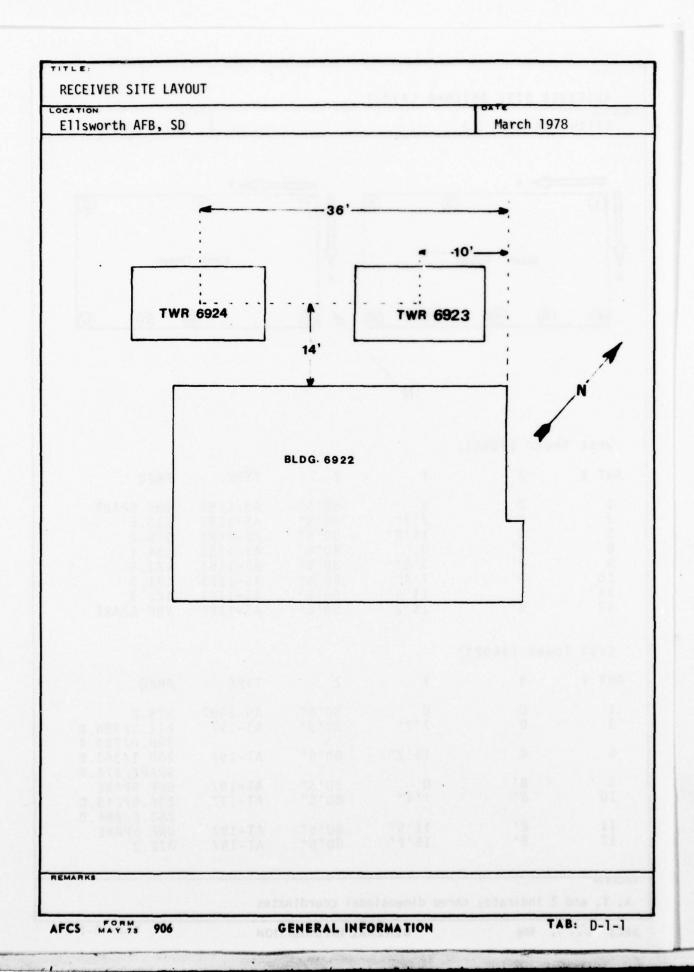
ALTITUDES FT. MSL 4000 5500 5000

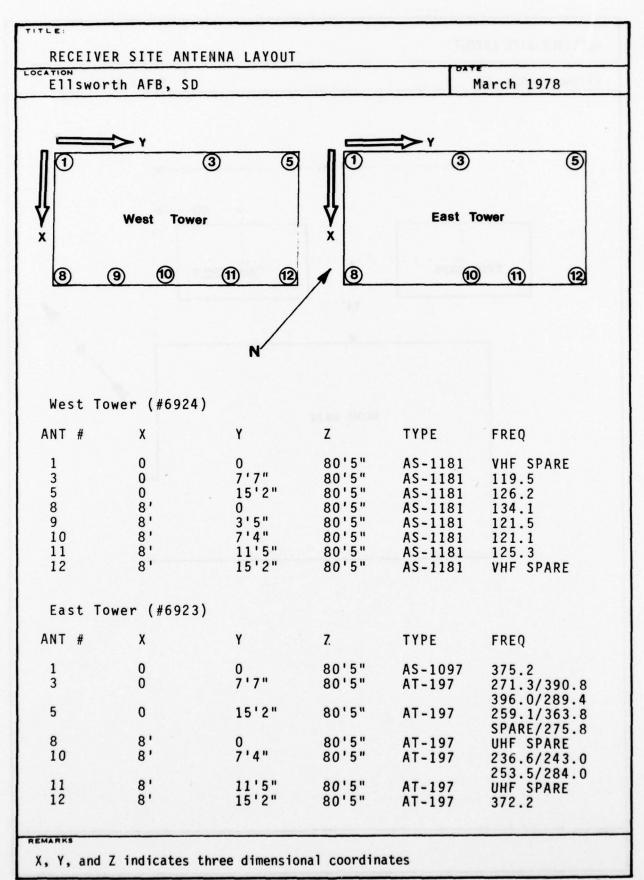
LINE OF SIGHT COVERAGE (RLS)



ELLSWORTH AFB RECEIVERS ANTENNA ELEVATION 3274 FT MSL SCALE: 1 INCH = 30 NM ORIENTED TO MAGNETIC NORTH VARIATION 13 DEG E

ALTITUDES FT. MSL 4000 5000





EQUIPMENT ANALYSIS SPECIFICATION LIST FREQUENCY: Self Explanatory 1. Equipment Type: Transmitters AN/GRT-21 and AN/GRT-22 (TO 31R2-2GRT-102) 2. Transmitter Serial Number: ----: Obtained from equipment 3. Percent of Modulation, 0 dBm Input: -: 90%+10% 4. Percent of Modulation, -15 dBm Input: 90%+10% 5. Percent of Modulation, +10 dBm Input: 90%+10% 6. Distortion:-----: 10% at lower limiting 15% at upper limiting 7. Frequency Accuracy Tolerance: ----: +0.001% with CR-143 crystal +0.002% with CR-75 crystal +0.0005% with freq synthesizer 8. Power Output:----- 10 Watts Minimum, Low power mode 50 Watts Minimum, High power mode 12.5 Watts max, 50 Watts forward 10. Transmission System VSWR:----: Normal operation at carrier power with VSWR not greater than 3 to 1 11. Coupler Loss:------ 2 dB Maximum (TO 31R1-2GR-142) CU-547 12. Antenna VSWR:------ 2:1 Maximum (TO 31R1-2GR-241) AS-1097 1.6:1 Maximum (TO 31R1-2GR-161) AT-197 Receiver Nomenclature: Receivers AN/GRR-23 and AN/GRR-24 (TO 31R2-2GRR-112) 14. Receiver Serial Number:----: Obtained from equipment 15. Frequency Accuracy Tolerance:----: +0.001% with CR-143 crystal +0.002% with CR-75 crystal +0.0005% with freq synthesizer 16. Sensitivity:----- 3 uv Maximum 17. Signal to Noise:-----: 10 dB with a 3 uv input 18. Sque1ch Threshold:----- 3 uv (TO 31R2-2GRR-116WC-1, 28 day inspection) 19. AGC Characteristics: : 3 dB Maximum variation with input signal of 6 uv to 1 v 20. Audio Output:----- +20 dBm 21. Distortion:-----: For frequencies 300, 1,500, and 3000 Hz with a 1 v input 10% maximum with 30% modulation 20% maximum with 90% modulation 22. Transmission System VSWR:-----: NSA (No Specifications Available) 23. Antenna VSWR:-----: 2:1 Maximum (TO 31R1-2GR-241) AS-1097 1.6:1 Maximum (TO 31R1-2GR-161) AT-197 24. Coupler Loss:----- 2 dB Maximum (TO 31R1-2GR-142) CU-547

REMARKS

TITLE EQUIPMENT ANALYSIS SPECIFICATION LIST FREQUENCY: Self Explanatory Equipment Type: Transmitter AN/GRT-18 (TO 31R2-GRT18-2) Transmitter Serial Number:----: Obtained from equipment 2. Percent of Modulation, 0 dBm Input:---: 90% Minimum 3. Percent of Modulation, -15 dBm Input:-: 90% Minimum 90% Minimum (50 Watt mode only). Percent of Modulation, +10 dBm Input:-: 5. 6. Distortion:----: NSA (No specification available). Frequency Accuracy Tolerance:----: +0.0014% of the assigned frequency. 7. Power Output:----: 10 Watts Minimum, Low power mode 50 Watts Minimum, High power mode. Reflected Power:----: 1.1 Maximum, 10 Watts forward 5.5 Maximum, 50 Watts forward 10. Transmission System VSWR:-----: 2:1 Maximum 11. Coupler Loss:----: NA 12. Antenna VSWR:------ 2:1 Maximum (TO 31R1-2UR-31) AS-1181 13. Receiver Nomenclature: Receiver AN/GRR-25 (TO 31R2-2GRR25-2) 14. Receiver Serial Number:-----: Obtained from equipment 15. Frequency Accuracy Tolerance:----: ±0.002% 16. Sensitivity:----- 5 uv 17. Signal to Noise:----: 10 dB with 5 uv input 18. Squelch Threshold:----: 3 uv maximum at max rf gain 19. AGC Control:----- 3 dB maximum variation with input signals of 15 uv to 1 v 20. Audio Output:-----: +10 to +30 dBm main audio (-10 to +10 dBm low level) 21. Distortion:----: 15% Maximum with a 1 v rf input signal modulated at 30% 25% Maximum with a 1 v rf input signal modulated at 90% 22. Transmission System VSWR:----: NSA 2:1 Maximum (TO 31R1-2UR-31) AS-1181 23. Antenna VSWR:----: 24. Coupler Loss:----:

REMARKS

Ellsworth AFB, SD					March 19		
FREQUENCY		39	0.8	363.	.8	23	6.6
I. TRANSMITTER NOMENCLA	TURE	AN/GR	T-22	AN/GR	Γ-22	AN/GRT	-22
2. SERIAL NUMBER		66	52	708	32	664	6
		INITIAL	ADJUSTED	INITIAL	ADJUSTED	INITIAL	ADJUSTE
MODULATION LEVEL	%	90		90		+100	90
4. LOWER LIMITING	%	90		90		90	
5. UPPER LIMITING	76	90		90		90	
. DISTORTION	7,	2.2		3.8		22	6
7. FREQUENCY ACCURACY	%	0.00		0.00		0.00	
RF POWER OUT	Watte	14	10	14	10	14	10
. COUPLER VSWR		N/A		N/A		1.1:1	
O. COUPLER LOSS	dB	N/A		N/A		2.6	
1. ANTENNA VSWR		1.1:1		1.1:1		1.2:1	
2. RECEIVER NOMENCLATUR	RE	AN/GR	R-24	AN/G	RR-24	AN/GRR-24	
3. SERIAL NUMBER		64	02	6	310	3051	
14. FREQUENCY ACCURACY	%	0.00		0.00	T	0.00	
15. SENSITIVITY	uv	1.7	1	3.4	1.5	9.1	
6. SIGNAL TO NOISE	dB	14.1		9.1	15.1	2.0	
7, SQUELCH THRESHOLD	UV	3.0		2.4	3.0	* NOTE	AND THE
8. AGC		1.6		1.2			
9. AUDIO OUT	dBm	20	1	16	20		
0. DISTORTION	7.	4.6		5.6			
1. COUPLER VSWR		1.3:1		1.1:1			
2. COUPLER LOSS	dВ	1.6		1.4	100	The state to	Acres 4
J. ANTENNA VSWR		1.5:1		1.1:1		Remodel	
REMARKS * Check was ter	minate	ed when re	eceiver wou	ld not ad	just to sp	ecification	ons.

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TAB: E-2-1

LOCATION					March 19	/8	
Ellsworth AFB, SD		125	.3	121	.5		
I. TRANSMITTER NOMENCLA	TURE	AN/GR		AN/GR		a termes	
2. SERIAL NUMBER		101		109			
		INITIAL	ADJUSTED	INITIAL	ADJUSTED	INITIAL	ADJUSTED
3. MODULATION LEVEL	7.	90		+100	90		
4. LOWER LIMITING	%	90		90			
5. UPPER LIMITING	%	90		90	90		
6. DISTORTION	%	5.4		4.8			
7. FREQUENCY ACCURACY	%	0.00		0.00			
RF POWER OUT 8. FORWARD	Watte	10		53	50		
9. COUPLER VSWR		N/A		N/A			
O. COUPLER LOSS	dB	N/A		N/A			
1. ANTENNA VSWR		1.2:1		1.4:1			
2. RECEIVER NOMENCLATUR	RE	AN/GR	R-25	AN/GRR-25		us-ragonie	
13. SERIAL NUMBER		82	2	83	6	and the second	
14. FREQUENCY ACCURACY	7,	+.00016		+.0004			
15. SENSITIVITY	uv	3	1.8	1.5			
16. SIGNAL TO NOISE	dĦ	10	20	15			
17. SQUELCH THRESHOLD	UV	3.3	3	3			
ie. AGC		2.6	1.8	1.1			
9. AUDIO OUT	dBm	-1	* 0	* 0			
20. DISTORTION	7.	20	17	5.1			
II. COUPLER VSWR		N/A		N/A			
Z. COUPLER LOSS	dВ	N/A		N/A			
. ANTENNA VSWR		1.4:1		1.6:1		1496	
REMARKS							

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TAB: E-2-2

TITLE AMPLIFIER DATA RAPCON March 1978 Type AM-4568/G Microphone Amplifier Serial Number PAR-1 PAR-2 ASR-1 ASR-2 ASR-3 CORD Position (dBm) Input Level -10 -10 -10 -10 -10 -10 (VRMS) Output Level 4.4 6.1 6.3 6.4 6.4 4.6 % Distortion (5% Max) 2.9 2.4 2.5 2.5 2.8 6.4 Noise Level (dBm) -57 -66 -62 -66 -66 -66 Input at Limiting (dBm) -13 -13 -13 -13 -13 -13 * * * * * Output at Limiting (dBm) * * NOTE: Unable to measure with available test equipment because of the higher than normal input level. Type AM-4568/G Microphone Amplifier Serial Number CL/DL Position MAINT (dBm) Input Level -10 -10 (VRMS) Output Level 6.2 6.8 (5% Max) % Distortion 2.3 2.1 -62 -63 (dBm) Noise Level (dBm) Input at Limiting -13 -13 Output at Limiting (dBm)

REMARKS

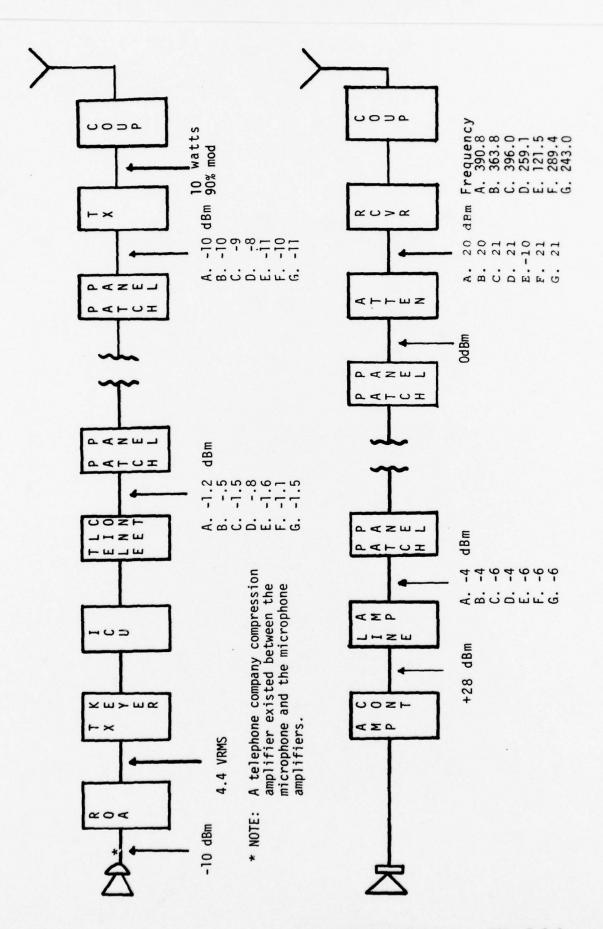
* NOTE:

Unable to measure with available test equipment because

of the higher than normal input level.

			Mar	ch 1978	
r	, , , , , , , , , , , , , , , , , , , 				,
298.4	125.3	259.1	243.0	121.5	121.5
) -5.8	-5	-5	-5	-5	-5
) 28	28	27	28	28	28
) 4	2.75	3.6	2.05	3.5	2.9
) -46	-45	-47	-46	-46	-46
	382				
r					
384.0	363.8	396.0	390.8	119.5	271.3
) -5	-5	-5	-5	-5	-5
) 28	27.5	28	28	27.5	27.6
2.5	2.9	2.6	2.8	3.5	3.7
) -46	-45	-42	-44	-45	-46
				13059	
	ikiah l				
134.1	UHF MULTI CH	VHF	VHF	UHF SPARE	
) -5				-1	
	28	28		28	
	3	3.8	3.5	3.2	J 92 6
3.1		And the second section is the second			
	-5.8 28 4 -46 -46 -7 384.0 -5 28 2.5 -46	298.4 125.3 1 -5.8 -5 28 28 1 4 2.75 1 -46 -45 28 27.5 28 27.5 28 27.5 29 -46 -45 20 -45	298.4 125.3 259.1 1 -5.8 -5 -5 28 28 27 4 2.75 3.6 -46 -45 -47 384.0 363.8 396.0 -5 -5 -5 28 27.5 28 2.5 2.9 2.6 -46 -45 -42	Marker 298.4 125.3 259.1 243.0 10 -5.8 -5 -5 -5 28 28 27 28 10 4 2.75 3.6 2.05 10 -46 -45 -47 -46 28 27.5 28 28 27 28 28 27.5 28 28 27.5 28 28 29 2.5 2.9 2.6 2.8 20 2.8 2.8 20 2.8 2.	March 1978 298.4

AM RADIO	COMMUNIC	ATIONS SY	STEM LOO	P ANALYSI	S	March	1978	
OCATION RAPCON		,						
I. FREQUENCY:		390.8	363.8	396.0	259.1	121.5	289.4	243.0
2. MIC AMP IN	dBm	-10	-10	-10	-10	-10	-10	-10
3. MIC AMP OUT	VRMS	4.4	4.4	4.4	4.4	4.4	4.4	4.4
4. NOISE FLOOR	dB Down	32	32	32	32	32	32	32
5. NOISE LEVEL	dBm	-66	-66	-66	-66	-66	-66	-66
6. CABLE IN	dBm	-1.2	5	-1.5	8	-1.6	-1.1	-1.5
7. NOISE FLOOR	dB Down	21	32	31	32	19	32	31
B. NOISE LEVEL	dBm	-30	-30	-30	-30	-28	-30	-28
9. CABLE OUT	dBm	-10	-10	-9	-8	-11	-10	-11
0. NOISE FLOOR	dB Down	37	37	38	38	37	37	37
1. NOISE LEVEL	dBm	-56	-65	-61	-59	-66	-64	-67
2. TRANSMITTER IN	dBm	-10	-10	-9	-8	-11	-10	-11
3. % MODULATION	*	90	90	90	90	90	90	90
4. POWER OUT	Watts	10	10	10	10	10	10	10
5. RECEIVER OUT	dBm	20	20	21	21	* 0	21	2
6. NOISE FLOOR	dB Down	31	28	30	29	22	28	29
7. NOISE LEVEL	dBm	-23	-27	-24	-24	-6	-25	-24
B. CABLE IN	dBm	0	0	0	0	0	0	(
9. NOISE FLOOR	dR Down	32	30	31	30	24	30	30
20. NOISE LEVEL	d₿m	-27	-30	-30	-28	-12	-30	-29
1. CABLE OUT	dBm	-4	-4	-6	-4	-6	-6	-6
2. NOISE FLOOR	dR Down	34	32	31	30	24	30	30
3. NOISE LEVEL	d₽m	-31	-35	-36	-32	-19	-36	-3
* Measureme								



LOOP TEST LINE LEVEL DIAGRAM (RAPCON)

Control Tower					Marc	h 1978	
Type AM-4571/G Line Ampl	ifier						
Frequency		236.6	126.2	253.5	396.0	243.0	289.4
Input Level	(dBm)	-4	-3	-3	-7	-7	-7
Output Level	(dBm)	26	26	26	26	26	26
% Distortion (5%	Max)	2.6	3.5	1.5	4.4	1.5	3.5
Noise Level	(dBm)	-30	-55	-57.5	-59	-54.5	-56
Type AM-4571/G Line Ampl	ifier						
Frequency	T	121.5	275.8				
Input Level	(dBm)	-7	-3				
		26	26				
Output Level	(GRW)						
	(dBm) Max)	1.75	1.95				
% Distortion (5%							
% Distortion (5%	Max)	1.75	1.95				
% Distortion (5% Noise Level	Max)	1.75	1.95				
% Distortion (5% Noise Level	Max)	1.75	1.95				
% Distortion (5% Noise Level Type AM-4571/G	Max)	1.75	1.95				
% Distortion (5% Noise Level Type AM-4571/G Input Level	Max) (dBm)	1.75	1.95				
% Distortion (5% Noise Level Type AM-4571/G Input Level Output Level	Max) (dBm)	1.75	1.95				

Control Tower				Mar	ch 1978	
Type AM-4571/G Speaker Ampli	Filou.					
Position	POS-T SPK-1	POS-T SPK-2	POS-T SPK-3	P0S-2 SPK-1	P0S-2 SPK-2	P0S-2 SPK-3
Input Level (dBm)		-39	-28	-41	-46	-47
	8	13.5	11	9.8	12	3.5
Output Level (dBm) % Distortion (5% Max)	3.6	6.9	1.4	8.2	3	2.7
Noise Level (dBm)	-55	-27	-60	-59	-57	-59
NOTSE LEVET (UDIII)				1 33	1 -5/	1 -55
Type AM-4571/G Line Amplifier						
Position	POS-3					
Input Level (dBm)	-42					
Output Level (dBm)	12					
% Distortion (5% Max)	5.2					
Noise Level (dBm)	-36					
Type AM-4571/G					3/3/5	T -
		<u> </u>	ļ			-
Input Level (dBm)		_				-
Output Level (dBm)				-		-
% Distortion (5% Max)		-				-
						10.1.11

Control Tower				Mare	ch 1978
ype AM-4568/G Microphone	Amplifier	(Initial)	e de de	
Serial Number	613	631	437	607	
Position	3	2	1	MAINT	
Input Level (dBm) -30	-30	-30	-30	
Output Level (VRMS) 1.1	2.8	3.2	1.7	
Distortion (5% Max) 1.4	1.3	1.6	1.1	
) -47	-65	-28	-41	
Noise Level (dBm					
Noise Level (dBm Input at Limiting (dBm		-41	-54	-59	
) -57	-41 -5	-54 -15	-59 -20	
Input at Limiting (dBm Output at Limiting (dBm) -57) -21 Amplifier	-5 (Adjuste	-15 d)	-20	
Input at Limiting (dBm Output at Limiting (dBm) -57) -21	-5	-15		
Input at Limiting (dBm Output at Limiting (dBm Type AM-4568/G Microphone Serial Number Position) -57) -21 Amplifier 613	-5 (Adjuste	-15 d)	-20	
Input at Limiting (dBm Output at Limiting (dBm Type AM-4568/G Microphone Serial Number Position Input Level (dBm) -57) -21 Amplifier 613 3) -30	-5 (Adjuste 631	-15 d)	607	
Input at Limiting (dBm Output at Limiting (dBm Type AM-4568/G Microphone Serial Number Position Input Level (dBm	Amplifier 613 3 3 1 1.4	-5 (Adjuste 631 2	-15 d) 437	-20 607 MAINT	
Input at Limiting (dBm Output at Limiting (dBm Type AM-4568/G Microphone Serial Number Position Input Level (dBm	Amplifier 613 3 3 1 1.4	-5 (Adjuste 631 2 -30	-15 d) 437 1 -30	-20 607 MAINT -30	
Input at Limiting (dBm Output at Limiting (dBm Type AM-4568/G Microphone Serial Number Position Input Level (dBm	Amplifier 613 3 3 -30 1.4 0.8	-5 (Adjuste 631 2 -30 1.6	-15 d) 437 1 -30 2.3	-20 607 MAINT -30 1.6	
Input at Limiting (dBm Output at Limiting (dBm Type AM-4568/G Microphone Serial Number Position Input Level (dBm Output Level (VRMS) % Distortion (5% Max	Amplifier 613 3 -30 1.4 0.8 -66	-5 (Adjuste 631 2 -30 1.6	-15 d) 437 1 -30 2.3 0.8	-20 607 MAINT -30 1.6 0.9	

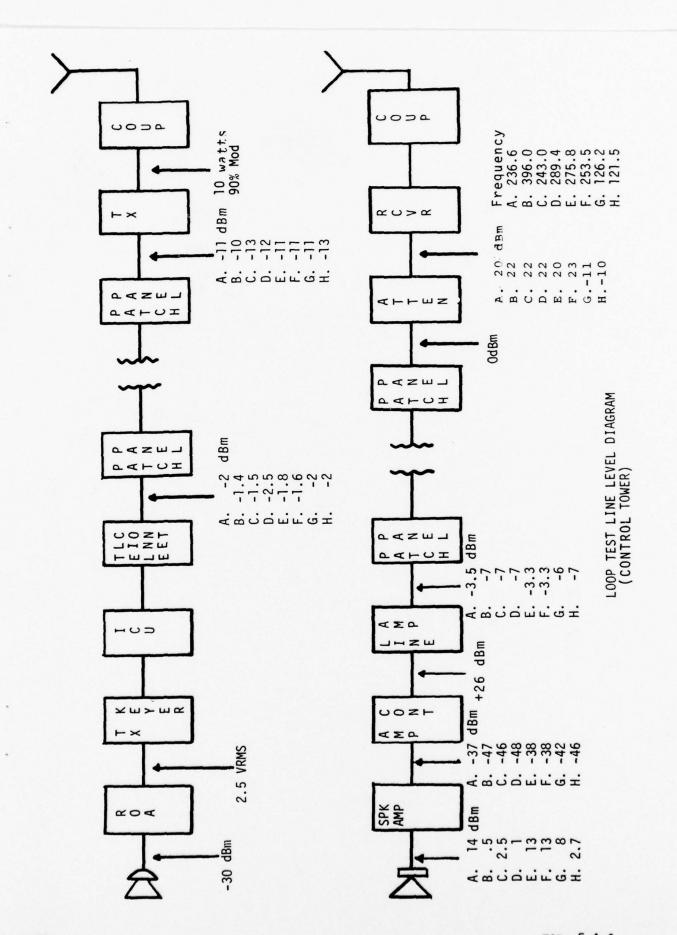
AM RADIO	COMMUNIC	ATIONS SY	STEM LOO	P ANALYS	IS	March	1978	
Control	Tower							·
1. FREQUENCY:		236.6	396.0	243.0	289.4	275.8	253.5	126.2
2. MIC AMP IN	dlim	-30	-30	-30	-30	-30	-30	-30
3. MIC AMP OUT	VRMS	2.5	2.5	2.5	2.5	2.5	2.5	2.5
4. NOISE FLOOR	dB Down	31	31	31	31	31	31	31
S. NOISE LEVEL	dBm	-78	-78	-78	-78	-78	-78	-78
5. CABLE IN	dBm	-2	-1.4	-1.5	-2.5	-1.8	-1.6	-2
7. NOISE FLOOR	dB Down	35	33.6	34	33.5	33.7	33.8	34
3. NOISE LEVEL	dBm	-39	-40	-40	-39	-39	-40	-40
. CABLE OUT	dBm	-11	-10	-13	-12	-11	-11	-11
. NOISE FLOOR	dB Down	34	35	34	33	35	34	34
1. NOISE LEVEL	₫Bm	-67	-55	-56	-54	-64	-67	-67
Z. TRANSMITTER IN	dBm	-11	-10	-13	-12	-11	-11	-11
3. % MODULATION	7.	90	90	90	90	90	90	90
4. POWER OUT	Watts	10	10	10	10	10	10	10
5. RECEIVER OUT	dRm	20	22	22	22	20	23	* 0
6. NOISE FLOOR	dB Down	22	23	21	18	25	15	23
7. NOISE LEVEL	dBm	-34	-33	-23	-23	-30	-24	-9
B. CABLE IN	dBm	0	0	0	0	0	0	0
9. NOISE FLOOR	dB Down	35	35	30	30	28	23	20
O. NOISE LEVEL	dBm	-38	-40	-30	-30	-33	-31	-12
1. CABLE OUT	d₿m	-3.5	-7	-7	-7	-3.3	-3.3	-6
2. NOISE FLOOR	dR Down	38	35	30	30	30	25	23
3. NOISE LEVEL	dBm	-40	-48	-37	-37	-35	-34	-15
4. SPEAKER AMP IN	dBm	-37	-47	-46	-48	-38	-38	-42
5. SPEAKER AMP OUT	dBm	14	.5	2.5	1	13	13	8
6. NOISE FLOOR	dB Down	19	28.5	25	27	17	17	15
7. NOISE LEVEL	dHni	-34	-37	-28	-30	-14	-28	-0.

AFCS JANTS 958 REVISED

AM RADIO	COMMUNIC	ATIONS SY	STEM LOC	P ANALYS	IS	March	1978	
LOCATION: Contro	ol Tower							
1. FREQUENCY:		121.5						
2. MIC AMP IN	d₽m	-30						
3. MIC AMP OUT	VRMS	2.5						
4. NOISE FLOOR	dB Down	31						
5. NOISE LEVEL	dBm	-78						
6. CABLE IN	dBm	-2						
7. NOISE FLOOR	dB Down	34						
8. NOISE LEVEL	dBm	-40						
9. CABLE OUT	dBm	-13						
10. NOISE FLOOR	dB Down	33						
11. NOISE LEVEL	dBm	-55						
12. TRANSMITTER IN	dBm	-13						
13. % MODULATION	OF Fre	90						
14. FOWER OUT	Watts	10						
15. RECEIVER OUT	dBm	* 0						
16. NOISE FLOOR	dB Down	21						
17. NOISE LEVEL	dBm	-8						
18. CABLE IN	dBm	0						
19. NOISE FLOOR	dB Down	25						
20. NOISE LEVEL	dBm	-12						
21. CABLE OUT	dBm	-7						
22. NOISE FLOOR	dB Down	24						
23. NOISE LEVEL	dBm	-20						
24. SPEAKER AMP IN	dBm	-46						
25. SPEAKER AMP OUT	dBm	2.7						
26. NOISE FLOOR	dB Down	25						
27. NOISE LEVEL	dBm	-9						
* Measureme	nt taker	at ou	tput of	ALTEC	amp.			

AFCS JAN 75 958 REVISED

TAB: E-4-5



LOCATION			EQUIPMENT	& SERIAL N	March	19/8	
	1 Tower						
CHECK	SPECIFICATIONS		PRIME POWER	R	5	TANDBY POW	ER
1. VISUAL			SAT			SAT	
Z. REGULA-		VOL	TAGE	Γ	VOL	TAGE	T
TOR INPUT		INITIAL	ADJUSTED	CURRENT	INITIAL	ADJUSTED	CURRENT
PHASE A		245		*	260		125
PHASE B		245		*	260		125
PHASE C		245		*	260		125
NEUTRAL							
3. REGULA- TOR OUTPUT							
PHASE A							
PHASE B							
PHASE C							
NEUTRAL							
	MANUFACTURER	TYPE			SERIAL NU		
GENERATOR	Caterpillar	A33TI	-	37B2937			
	150KW	60 H:					
AUTOMATIC	MANUFACTURER	TYPE	•		CHANGEO	ER INTERVAL	
CHANGEOVER	Westinghouse	Auto			8 sec		
		VOLTAGE REGUL	ATOR RESPO	NSE			
VOLTAGE REGULATOR	SPECIFICATION	AS FOUND	MANU	ADJUSTED	TO:		TO A DJUST
PHASE A				-		+	
PHASE B							
PHASE C			<i>y</i> 12				
EQUIPMENT G	ROUNDING:						
REMARKS							
* NOTE:	Commerical buss w	as too large	to measur	re with a	vailable	e equipmen	nt.

SAT TAGE CURREN		
SAT TAGE CURREN		
TAGE CURREN		
ADJUSTED CURREN		
177		
175		
175		
2539		
116KW		
C		
TIME TO ADJUS		

	A. C. F	OWER			March	1978	
LOCATION	6.1		EQUIPMENT	SERIAL	UMBER		
Transmitt	er Site						
CHECK	SPECIFICATIONS		PRIME POWER	?	S	TANDBY POW	ER
NSPECTION			SAT			SAT	
OR INPUT		VOL INITIA L	TAGE ADJUSTED	CURRENT	VOL	ADJUSTED	CURRENT
PHASE A		123		17.5	130		18
PHASE B		124		15	130		18
PHASE C		125		16 130			17
NEUTRAL							
3. REGULA-							
PHASE A							
PHASE B							
PHASE C							
NEUTRAL							
SENERATOR -	Buda	80C12	90		65697	MBER	
	30KW	60 Hz			7KW		
AUTOMATIC CHANGEOVER	Zenith	Auto		Markini.	* NOTE	ER INTERVAL	
VOLTAGE	T	VOLTAGE REGUL	ATOR RESPO	ADJUSTE	то:	1	
REGULATOR	SPECIFICATION	AS FOUND	MANUA	ALLY	AUTOMATIC		TO ADJUST
PHASE A							
PHASE B							
PHASE C							
EQUIPMENT GR	OUNDING:		1				
REMARKS:							
* NOTE:	Automatic changeo	ver unit was	awaiting	parts a	nd could	not be cl	hecked.

	A. C.	March 1978							
LOCATION		MUMBER 1970							
Receiver	Site								
CHECK	SPECIFICATIONS			PRIME POWER			STANDBY POWER		
1. VISUAL			SAT			SAT			
2. REGULA-			VOLTAGE CURRENT		VOLTAGE		CURRENT		
TOR INPUT			INITIAL	ADJUSTED		INITIAL	ADJUSTED		
PHASE A			120		5.5	122		5.7	
PHASE B									
PHASE C									
NEUTRAL									
3. REGULA- TOR OUTPUT									
PHASE A	<u>~</u>								
PHASE B									
PHASE C									
NEUTRAL									
	Consolidated Die	EMU-11S			SERIAL NUMBER 65-0056				
GENERATOR	5KW	FREQUENCY 60 Hz			.8KW				
	AUTOMATIC CHANGEOVER Consolidated Dies		sel Auto			20 Sec			
				ATOR RESPO	NSE				
VOLTAGE REGULATOR	SPECIFICATION		AS FOUND	MANUALLY		AUTOMATIC	TIME	TIME TO ADJUST	
PHASE A									
				+			1-		
PHASE B				-					
PHASE C									
EQUIPMENT G	ROUNDING								
REMARKS:									

RSL MEASUREMENT FLIGHT PROFILE

Ellsworth AFB, SD

March 1978

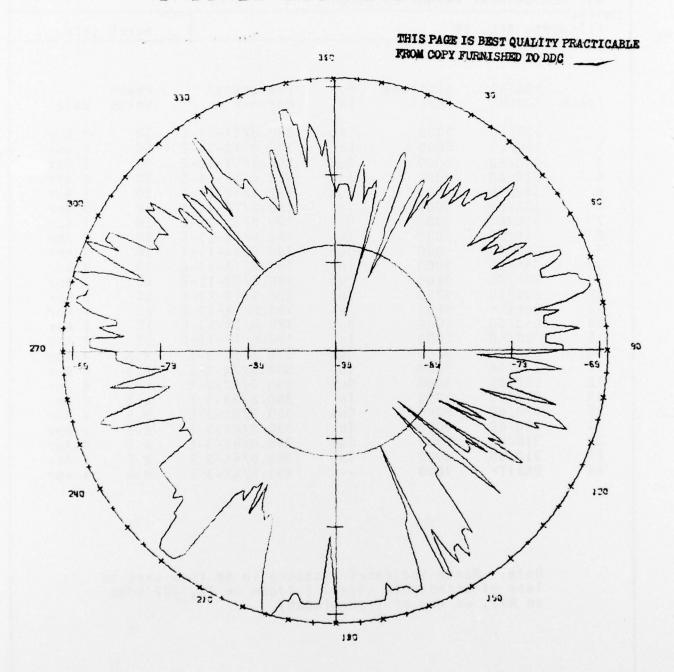
Track	Radial/ Range	Altitude (MSL)	Out In	Frequency/ Antenna	Power Watts	Date
1	130/60	5000	Out	390.8/TX-11-E	10	3 Apr
2	130/60	5000	In	390.8/TX-11-E	10	3 Apr
3	310/60	6000	Out	390.8/TX-11-E	10	3 Apr
4	310/60	6000	In	390.8/TX-11-E	10	3 Apr
3 4 5 6 7	130/60	9000	Out	390.8/TX-11-E	10	3 Apr
6	130/60	9000	In	390.8/TX-11-E	10	3 Apr
7	310/60	9000	Out	390.8/TX-11-E	10	3 Apr
8	310/60	9000	In	390.8/TX-11-E	10	3 Apr
9	050/60	9000	Out	390.8/TX-11-E	10	3 Apr
10	050/60	9000	In	390.8/TX-11-E	10	3 Apr
11	230/50	9200	Out	390.8/TX-11-E	10	3 Apr
12	230/50	9200	In	390.8/TX-11-E	10	3 Apr
13	ORBIT*	9600		390.8/TX-11-E	10	4 Apr
14	050/60	5000	Out	390.8/TX-11-E	10	4 Apr
15	050/60	5000	In	390.8/TX-11-E	10	4 Apr
16	210/55	9200	Out	390.8/RX-3-E	8.5	4 Apr
17	210/56	9200	In	390.8/RX-3-E	8.5	4 Apr
18	050/60	9000	Out	390.8/RX-3-E	8.5	4 Apr
19	050/60	9000	In	390.8/RX-3-E	8.5	4 Apr
20	130/60	9000	Out	390.8/RX-3-E	8.5	4 Apr
21	130/60	9000	In	390.8/RX-3-E	8.5	4 Apr
22	310/60	9000	Out	390.8/RX-3-E	8.5	4 Apr
23	310/60	9000	In	390.8/RX-3-E	8.5	4 Apr
24	ORBIT*	9600		390.8/RX-3-E	8.5	4 Apr

Note: Range indicates distance in NM from base to loss of acceptable signal (-93dBm on TX, -97.5dBm on RX), or end of track (60NM).

REMARKS

^{*} NOTE: Indicates a radius of 30 NM.

MEASURED SIGNAL STRENGTH



ELLSWORTH AFB RECEIVER SITE AN/GRT-22 RANGE 30 NM. ALTITUDE 9600 FT. MSL FREQUENCY 390.80 MHZ

VARIATION 13 DEGREES EAST

SCALE 1 INCH = 10 DB

ORIENTED TO MAGNETIC NORTH

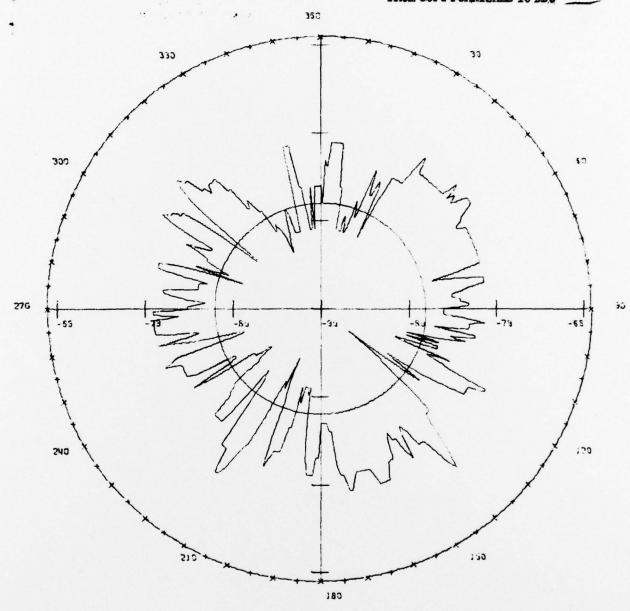
-87 DBM = 10 UVOLTS

4 APR 78

TAB: F-2-1

MEASURED SIGNAL STRENGTH

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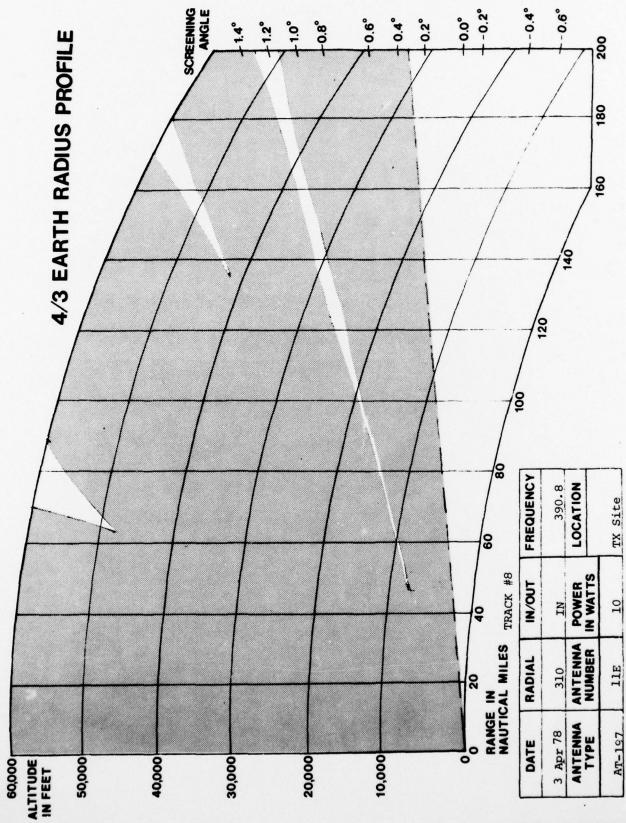


ELLSWORTH AFB RECEIVER SITE AN/GRR-24 RANGE 30 NM. ALTITUDE 9600 FT. MSL FREQUENCY 390.80 MHZ

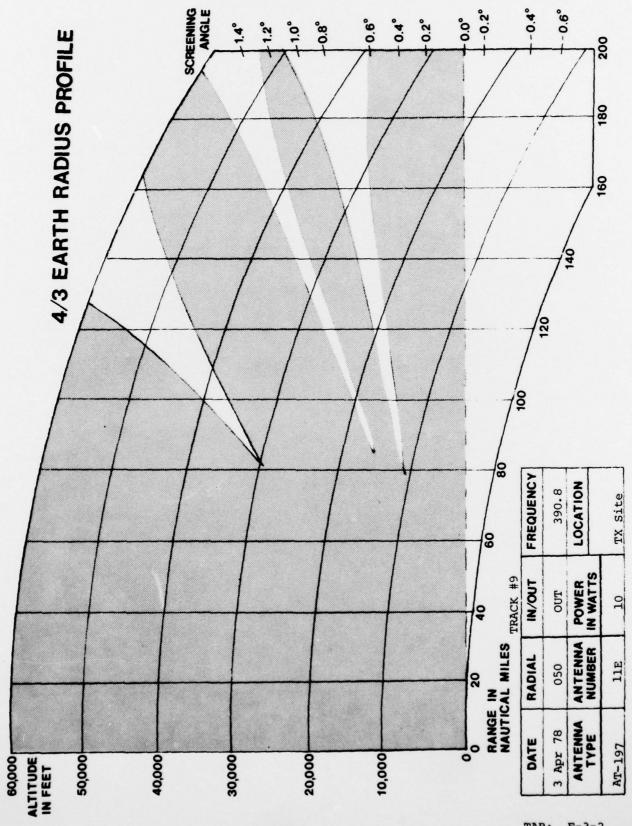
VARIATION 13 DEGREES EAST SCALE 1 INCH = 10 DB ORJENTED TO MAGNETIC NORTH -87 DBM = 10 UVOLTS 4 APR 77

TAB: F-2-2

RADIATION PATTERN (-93.0 dBm REFERENCE)

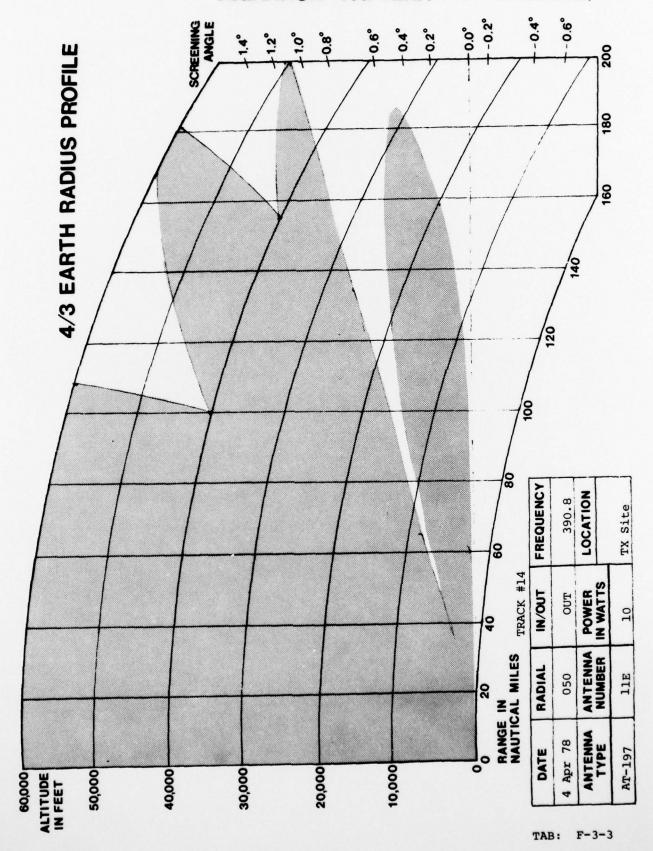


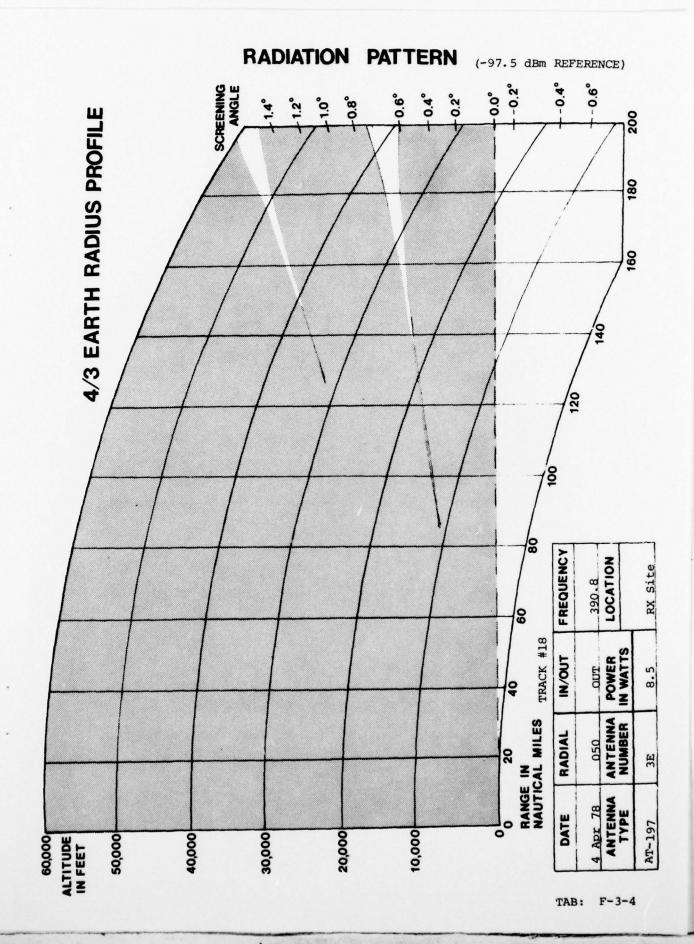
RADIATION PATTERN (-93.0 dbm REFERENCE)



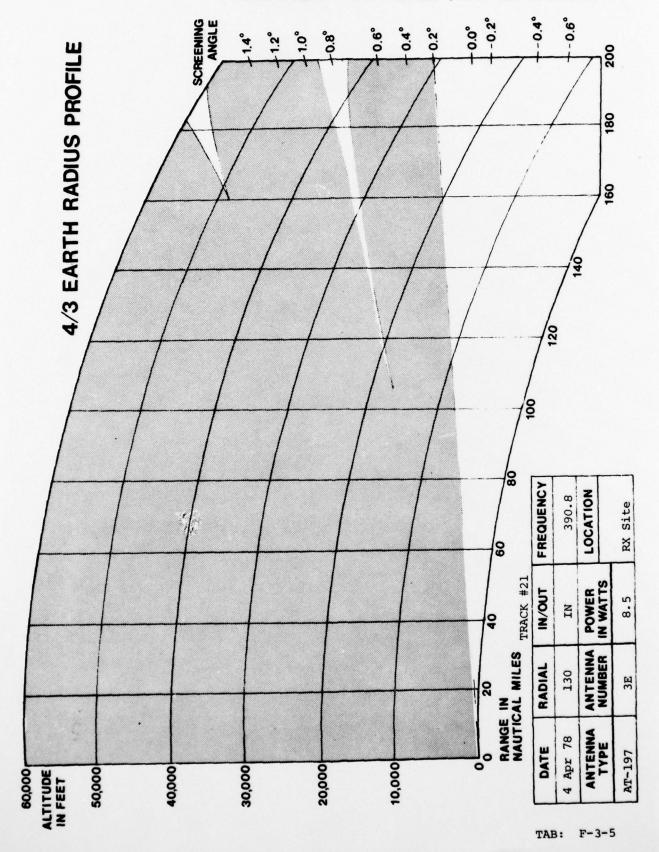
TAB: F-3-2

RADIATION PATTERN (-93.0 dbm REFERENCE)



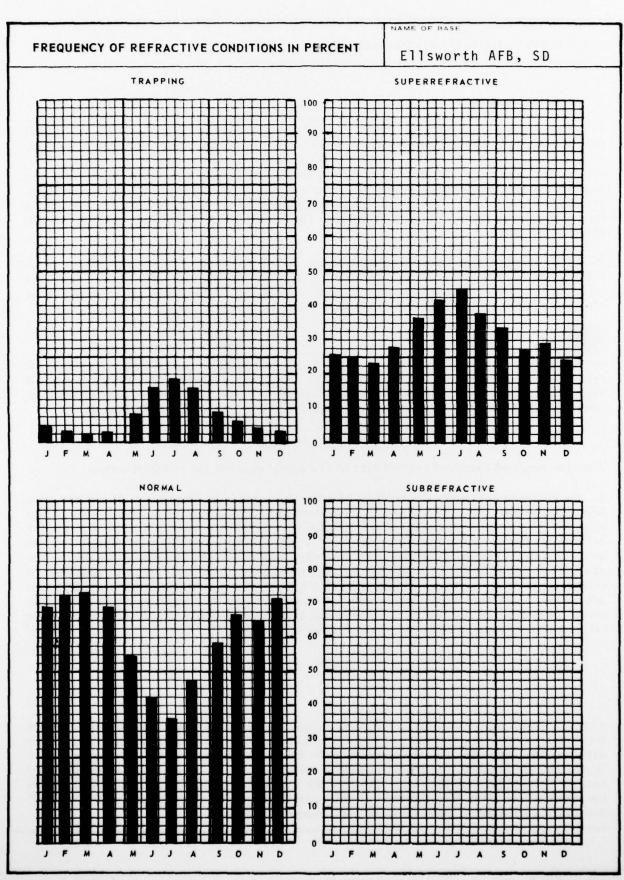


RADIATION PATTERN (-97.5 dBm REFERENCE)



RADIATION PATTERN (-97.5 dBm REFERENCE) -0.4 -0.6° SCREENING -0.2° -0.0° 0.40 -0.2° 0.0 1.00 **.**0.8° 1.20 4/3 EARTH RADIUS PROFILE 200 180 160 5 FREQUENCY LOCATION 390.8 RX Site TRACK #22 POWER IN WATTS IN/OUT OUT 8.5 ANTENNA RANGE IN NAUTICAL MILES RADIAL 310 ANTENNA Apr 78 DATE AT-197 10,000 30,000 20,000 40,000 ALTITUDE IN FEET 50,000 60,000

TAB: F-3-6



- 1. The bending or refraction of electromagnetic energy as it passes through the air occurs because of the structure of the troposphere. Energy propagated through a vacuum would travel in a straight line. Similarly, energy transmitted through any gas (or liquid) that is uniform in density perpendicular to the direction in which the energy is traveling, will follow a straight line path. However, due to the physical characteristics of the troposphere, the density of the troposphere decreases with increasing height. Therefore, the front of energy transmitted at low elevation angles will be subject to refractive bending. Usually, the top of the wave front will move faster than the bottom, since the density of the atmosphere decreases with height. The result is a downward bending of the transmitted energy.
- 2. The number that describes the relative speed of propagation in any substance is referred to as the index of refraction (n). It is defined as the ratio of the speed of propagation of electromagnetic energy in a vacuum (c) to the speed of propagation of electromagnetic energy in the medium in question (v):

$$n = \frac{c}{v}$$

Within the wavelength band from 1 cm (30 GHz) to 10 meters (30 MHz), the index of refraction does not change appreciably as the frequency changes. The typical range of values of n at sea level is from 1.000250 to 1.000450. Since these numbers are difficult to work with, a "scaled-up" quantity called refractivity (N) is used, and is defined as

$$N = (n - 1) \cdot 10^6$$

Thus the range of values of refractivity at sea level becomes 250 to 450 N-units.

3. As mentioned earlier, the bending of energy is caused by the change in density with height in the air. Since the speed of propagation of energy is related to the density of the air, and the refractivity (N) is related to the speed of propagation of energy (by definition), then refractivity in the troposphere is directly related to the density of the air. Therefore, the bending of electromagnetic energy may be thought of as due to the change of refractivity with height in the troposphere, or the vertical gradient of refractivity. It is important to note that it is not the value of N at a particular point that determines refraction but it is the gradient of refractivity that must be considered. The refractivity may be related to the meterorological variables of pressure (p), temperature (T), and water vapor pressure (e) by the following equation:

$$N = \frac{Ap}{T} + \frac{Be}{T^2}$$

where A and B are constants. The normal rapid decrease of p and e with height in the troposphere leads to a decrease of N with height. Temperature usually decreases slowly with height, and this has an opposite effect on the change of N. In the so-called "standard" atmosphere, the result is that N will decrease by about 12 N-units per 1000 feet of altitude through the lower levels of the troposphere, and 6 N-units per 1000 feet in the upper levels. It is this decrease of refractivity with height that leads to the "normal" downward curvature, or refraction, of electromagnetic energy.

- 4. In the "real" troposphere all is not so simple. The temperature and water vapor pressure may vary in any manner, while atmospheric pressure will continue to decrease with height. This seemingly random variation of the meteorological terms will lead to unusual changes in refractivity with height. Refractivity may decrease more than in the "standard" troposphere, causing more pronounced bending of electromagnetic energy. On the other hand, refractivity may actually increase with height, which may result in an upward curvature of a radio/radar beam (opposite the curvature of the earth). The propagation of electromagnetic energy along a path that is different from the usual or expected path is known as "anomalous propagation" (AP). The refraction that results under various AP conditions is referred to as either subrefraction, superrefraction, or trapping (ducting). These refractive conditions, the effects on electromagnetic energy presented as a single ray, and the gradients of refractivity that may cause them are defined below:
- a. Subrefraction: Ray curvature is upward. Radio/radar ranges are significantly reduced. The occurrence is quite rare. The gradient of refractivity is equal to or greater than 0 N-units/1000 feet (average "standard" value is 12 N-units/1000 feet).
- b. Normal refraction: Ray curvature is downward but not as much as the curvature of the earth. Radio/radar performance is generally undisturbed, and the occurrence is frequent. The gradient of refractivity is less than 0 N-units/100 feet and greater than 24 N-units/100 feet.
- c. Superrefraction: Ray curvature is downward, more sharply than normal, but not as much as the curvature of the earth's surface. Radio/radar ranges may be significantly extended; the occurrence is frequent. The gradient of refractivity is greater than -48 N-units/100 feet and less than or equal to -24 N-units/1000 feet.
- d. Trapping: Extreme superrefraction, with downward curvature equal to or greater than the curvature of the earth's surface. Radio/radar performance is greatly disturbed, ranges are greatly extended, holes in coverage may appear; occurrence is not normally frequent. The gradient of refractivity is less than or equal to -48 N-units/1000 feet.
- 5. For an understanding of refractive effects on the system being evaluated, refer to AFCS Pamphlet 100-79.